

1 Theoretical approach to the scale effects of an OWC device

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7 **Abstract**

This research deals with the dynamic similarity problem for Oscillating Water Column (OWC) devices, for which air is the fluid that is subject to thermodynamic transformations in the inhalation/exhalation phases. Based on the differential problem, both linearised and full-nonlinear, the scale ratios satisfying similarity are calculated, with specific reference to the case where constraints are present on some of these scale ratios. The paper proceeds to identify the numerous processes of a turbulent interface that scales differently between model and prototype. With the aim of bringing to front the influence of the scale effects on featured aspects of the thermodynamic process involved, it is proposed that a non-equilibrium thermodynamics approach can be more comprehensive and representative not only of transformations, but also of scaling. The study reveals that in the case of OWC thermodynamics, non-equilibrium states which would be less evident in scaled model, would become more relevant as the scale is increased towards the size of the prototype, with consequences on performance.

8 *Keywords:* wave energy, oscillating water column, thermodynamics, scale effects,  
9 similarity, polytropic process

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10 **1. Introduction**

11 The ocean dynamics appears as a potential source of renewable energy for primary  
12 conversion with an essentially permanent availability. Estimates suggests  $\sim 10^7$  MW of  
13 off-shore available wave power over the coasts worldwide ([17], [7]), representing  $\sim 34\%$   
14 of the total primary conversion in Europe, [67]. In a world climate change scenario, it  
15 is a priority to develop technologies that allow to use the ocean resource for primary  
16 conversion as a complement/replacement of fossil fuels.

17  
18 Nowadays, the Oscillating Water Column (hereinafter OWC) is the most remarkable  
19 wave energy converter device. One of its most important features is the fact that the  
20 only mobile element is the turbine, which simplifies the design and the costs of the device  
21 ([20, 54]). Several full-scale plants have been build: Mutriku (Spain), Pico (Portugal),

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22 Port Kembla (Australia), and Niigata(Japan) among others. Nevertheless, different  
23 targets must be achieved to make this technology a real alternative, (i) to minimize  
24 the installation and deployment costs, (ii) to find technical solutions that make it an  
25 attractive framework for benchmarking, (iii) to find technical solutions to satisfy the end  
26 customers, or to get the social acceptance ([3, 27, 64, 49, 50, 51, 28]).

27  
28 Different research lines have been focused on the development of the OWC devices.  
29 The theoretical performance of the OWC has been studied by solving analytically the  
30 radiation–diffraction problem ([10, 58, 11]); other research have focused on the power  
31 take-off (PTO) control and performance efficiency and management ([2, 16, 14, 21]).  
32 Some authors have studied the boundary conditions of the radiation-diffraction problem  
33 ([35, 39, 38]), as well as the implementation of the OWC embedded in vertical breakwaters  
34 ([45, 19]), the interaction between the OWC and the seabed and its long–time response  
35 ([55, 56, 41, 42]), the development of the floating OWCs to eliminate the problems  
36 associated with the installation in deep waters ([24, 57]), and the development of a  
37 new concept of turbine ([22]). Numerical simulations and experimental tests have been  
38 carried out to improve the knowledge about the OWC devices under controlled conditions,  
39 impossible to achieve otherwise, such as the hydrodynamic and aerodynamic coupling  
40 ([60]), the non–linear considerations to increase the OWC efficiency ([36]), and the  
41 implementation of the Actuator Disk Model for turbine simulations (see, e.g., [47]). The  
42 problem of physical and numerical modelling of the turbine is still open and discussed.  
43 In addition to the traditional Wells and impulse turbines, the construction and adoption  
44 of axial impulse turbines with design criteria already widely used in turbomachinery  
45 is proposed ([4]). In particular, wave-to-wire modelling has been conceived, with a  
46 holistic approach that includes turbine control ([5]). This means that the overall model  
47 includes three sections: (i) a primary converter model to convert wave motion into  
48 pressure fluctuations in the OWC; (ii) a secondary converter model to convert air pressure  
49 fluctuations into torque at the turbine axis; (iii) a tertiary converter model to convert  
50 torque at the turbine axis into electrical energy generated by an electric generator. The  
51 model is then used to optimize performance in relation to the plant location, as many  
52 parameters need to be tuned to maximize performances, including average annual wave  
53 statistics and inter-annual variability of the meteorological climate. All this in a context  
54 where the average price per unit of energy produced is still uncompetitive with many  
55 other energy sources, not least because of maintenance costs in an adverse environment;  
56 the overall yield is almost always in the single-digit percentage range. In order to resolve  
57 problems related to the social acceptance, some authors have studied the combination  
58 of OWC and hydrogen electrolysis for wave energy extraction and criteria management  
59 ([29]).

60  
61 One key factor in the OWC performance is the thermodynamics of the air chamber.  
62 The efficiency of the device is closely related to the nature of the gas inside the chamber  
63 and its compression/expansion cycles, which results in a polytropic transformation. The  
64 application of the First Law of Thermodynamics to the open system of the air chamber  
65 can be done by *transforming* the open system into a close one ([31]). That process  
66 has been successfully studied under the assumption of the isentropic process of an ideal  
67 gas ([16, 59, 68]). Nevertheless, the implementation of the real gas model using the  
68 virial Kammerlingh–Onnes expansion helps to justify the low OWC efficiency values

69 ([23, 53, 66, 63]). This fact is tested under experimental tests and numerical solutions  
70 of the radiation-diffraction problem with real gas implementation ([44, 43, 40]). Other  
71 researches point that the process is not totally adiabatic ([46]), as well as highlight the  
72 role played by the turbine as a restraint of the thermodynamic system, affecting the  
73 pneumatic efficiency ([46, 25]).

74  
75 Although numerical simulations can provide information about the OWC perfor-  
76 mance, there are limitations when some specific features are implemented, such as  
77 the combination of dry air and moisture, the real gas model or the non-adiabatic  
78 process. Those problems can be solved using experimental tests which allow to repro-  
79 duce situations in which different parameters can be controlled, impossible to control  
80 otherwise. The experimental tests must be done in a reduced scale to minimize the cost  
81 of the test and to adapt them to the space available in the laboratories. In this sense,  
82 the dimensional analysis allows to establish the scale between the real model and the  
83 prototype in order to ensure the prototype performance in the same way as the full-scale  
84 model. Nevertheless, in this scale transformation new problems can appear related to  
85 scale effects. These effects can be reduced or well quantified applying the dimensional  
86 analysis ([34]).

87  
88 The scale effect affecting the OWC devices have been comprehensively studied by  
89 several authors, like [65, 8] among others. Traditionally, the scale factor has been  
90 calculated using the Froude similarity ([37, 13, 15]), with the particularity that the  
91 volume scale is  $r_{\rho_w} \lambda^2$ , where  $\lambda$  is the length scale factor, and  $r_{\rho_w}$  is the water density  
92 ratio ( $\rho_{scaled\ model}/\rho_{prototype}$ ). Nevertheless, some authors have considered the problem  
93 separated into two parts: the hydrodynamic problem governed by the Froude similarity,  
94 and the aerodynamic problem governed by the Mach similarity. This solution leads to  
95 consider the same height of the chamber both in the model as in the prototype ([65]).  
96 Both peculiarities —the volume scale  $V \sim \lambda^2$  and the constant height of the chamber—  
97 lead to the use of a rigid-walled bellow of air in the model ([12]).

98  
99 The non-dimensional focusing on the OWC problem has been carried out under  
100 several scopes, namely hydrodynamic, aerodynamic and even thermodynamic ([65, 12,  
101 48, 44, 46], among others). All in all, scale effects eventually present in the observations  
102 of model tests are difficult to identify and isolate from previous results, partly due to  
103 the fact that there is not enough information to compare. It would be desirable to have  
104 real data about thermodynamics variables to compare, but those variables have not been  
105 recorded in real-scale prototypes or they are not available, as far as the Authors in this  
106 research have been concerned.

107  
108 The objective of this research is to develop a theoretical framework for a com-  
109 prehensive understanding of the possible scale effects on the OWC performance and  
110 their interrelations, ultimately leading to a reliable estimate of the OWC efficiency.  
111 The dimensional analysis will focus on how the scale effects can affect to fundamental  
112 governing hydrodynamic and thermodynamic variables. In particular, the study points  
113 to the scale effect on the polytropic exponent, which determines the nature of the system  
114 process equation defining the air compression and expansion processes inside the OWC  
115 chamber.

116

117 This paper is organized as follows. First, the differential problem is introduced,  
118 describing the thermodynamic process in the chamber, providing similarity rules for the  
119 linearised problem –section § 3.1– and for the full-nonlinear problem –section § 3.2–.  
120 Section § 4.1 describes the similarity rules for the water side and section § 4.2 describes  
121 the scaling of the turbines, as alternative to hole and porous layer usually adopted for  
122 simulating quadratic and linear characteristics of the turbine. Section § 4.3 analyzes  
123 the scale effects due to non respecting Reynolds, Weber and Mach similarity, but only  
124 Froude similarity. Section § 5 describes an instability analysis for the polytropic exponent.  
125 Finally, discussion and conclusion sections bring to front possible links between governing  
126 variables affected by scale effect.

## 127 2. Reach and novelty of the research

128 This research focuses on the scale effects in the thermodynamic compression–expansion  
129 process from a primary theoretical approach, to be later implemented in experimental  
130 observation. As far as the Authors are concerned, no research is available in terms of scale  
131 problems of the whole OWC wave-to-wire setup. This research is intended brings to front  
132 a theoretical approach to the scaled OWC thermodynamics, as a reference to be later  
133 observed in experimental testing. All in all, there are prior experiences by other authors  
134 —Falcão & Henriques [12]— that reveal that the approach makes sense. This paper  
135 helps to understand how the thermodynamic scaling requires a different adjustment as  
136 the standard scaling applied to other process involved in OWC performance. In fact,  
137 the accuracy of thermodynamic processes experimentally simulated increases downward  
138 –from full scale to model-, due to the minimization of transient states between equilibrium  
139 states, while the rest of process involved in OWC performance gain in accuracy upward  
140 –from model to full scale. In addition, if a research focus in the wave-to-wire model,  
141 the accuracy that can be reached in other aspects, like the thermodynamics processes  
142 for example, will be lower, and vice versa. So it seems not totally feasible to get a great  
143 accuracy in all the different aspects of the whole process.

144  
145 On the one hand, the study of the OWC chamber must be extended as a whole  
146 to the full problem. Otherwise, coupling different parts of the process that are in  
147 turn interrelated, might lead to a mismatched conclusions. However, in experience of  
148 the Authors in this research, a complete approach to the OWC system performance in  
149 which radiation-diffraction, turbine performance, power extraction and generator-to-grid  
150 connection would be otherwise a somewhat unreachable task. As far as the Authors  
151 are concerned, approaching the problem from different points allows to focus on specific  
152 aspects, yet to be clearly understood prior to build up a complete view. In that sense,  
153 state of the art reveals that this has been the way in which wave power extraction in gen-  
154 eral and OWC technology in particular have been studied. The theoretical formulation  
155 of the radiation-diffraction problem was conducted assuming pressure-air flow coupling  
156 based on a linearized isentropic relation —Evans [10], Sarmiento & Falcão [58], Martins-  
157 Rivas & Mei [39, 38]—. Once the theoretical basis was settled, different research lines  
158 were devoted to advance separately on specific features of OWC performance, including  
159 turbine performance, turbine damping, chamber performance in which simplifications  
160 were assumed such as the replacement of the turbine by an orifice or an actuator disk

161 model, wave action simulated by a piston type motion, etc.

162  
163 Some authors —Henriques *et al.* [26], Ciappi *et al.* [5]— have successfully developed  
164 a complete wave-to-wire model that connects the wave action through the different  
165 transformations stages to the final connection to the grid. Even in the case, some  
166 simplifications have to be assumed when coming up with the pressure-air flow coupling,  
167 such as adiabatic process and replacement of the turbine by an actuator disk model, given  
168 the difficulties to represent the turbine performance. While that type of model provides  
169 with a really accurate approach to the complete process, there remain specific aspects  
170 that require a comprehensive yet detailed focusing. This is the case of thermodynamic  
171 properties bound to scale effects when dealing with experimental testing.

### 172 3. Dynamics and Thermodynamics of OWC

173 Let consider the system consisting of the air chamber in which the internal volume  
174 changes periodically as a result of wave action. Let us assume, for simplicity, that the  
175 air chamber is vertical cylindrical with a homogeneous cross-sectional area  $A_c$ . The  
176 turbine is schematised as having a pressure drop proportional to the velocity of the  
177 air flow exchanged with the external environment, or proportional to the square of the  
178 velocity, to schematise a Wells-type or impulsive-type turbine, respectively. The mass  
179 conservation equation reads:

$$\frac{dm}{dt} = -Q_m, \quad (1)$$

180 where  $m(t)$  is the instantaneous mass of the gas (air) in the chamber and  $Q_m$  is the mass  
181 flowrate exchanged with the ambient through the PTO cross-section. Since  $m = \rho_c V$ ,  
182 where  $\rho_c$  is the gas density in the chamber and  $V$  is the volume of the chamber, eq.(1)  
183 can be written as

$$\rho_c \frac{dV}{dt} + V \frac{d\rho_c}{dt} = -Q_m. \quad (2)$$

184 The mass flowrate  $Q_m$  can be expressed as

$$\begin{cases} Q_m = \rho_c A_{pto} v, & \text{during exhalation,} \\ Q_m = \rho_a A_{pto} v, & \text{during inhalation,} \end{cases} \quad (3)$$

185 where  $A_{pto}$  is the cross-section area of the PTO device and  $v$  is the space average air  
186 velocity on  $A_{pto}$ , positive during exhalation and negative during inhalation. Here  $\rho_a$  is  
187 the ambient air density. It is necessary to analyse the process of exhalation and that of  
188 inhalation separately, since in the former, air escapes from the chamber with a density  
189 greater than that at atmospheric pressure; in the latter, the density of the air flow is  
190 equal to that at atmospheric pressure.

191  
192 The system evolution through equilibrium states addresses the polytropic process  
193 equation in its most general form

$$\frac{p_c}{\rho_c^n} = \text{constant}, \quad (4)$$

194 where  $n$  is the polytropic exponent. We assume that the air behaves as an ideal gas,  
 195 which implies that  $n = \gamma$ , where  $\gamma = 1.4$  for air in adiabatic —or isentropic in the  
 196 case of adiabatic and reversible— transformations, and  $p_c$  is the absolute pressure in the  
 197 chamber. The volume of the air in the chamber changes in time because part of it is  
 198 periodically invaded by the water, hence

$$V = A_c (h_0 - \eta(t)), \quad (5)$$

199 where  $h_0$  is the chamber height at rest and  $\eta(t)$  is the instantaneous cross-section average  
 200 water level in the chamber. As a first approach, we are neglecting the water column  
 201 dynamics in the chamber, which results from the interaction between the OWC and the  
 202 external wave field.

### 203 3.1. Similarity rules for a linear characteristic of the PTO device

For a Wells turbine, as a first approximation we assume the following linear relation  
 between pressure drop and air velocity:

$$p_c - p_a = K_1 v, \quad (6)$$

where  $K_1$  with dimension  $[K_1] = ML^{-2}T^{-1}$  is the air flow damping coefficient, assumed  
 invariant during exhalation/inhalation, and  $p_a$  is the absolute atmospheric pressure. In  
 order to compare our analysis with previous analyses, we first express all the terms in  
 eq.(2) as a function of the absolute pressure, obtaining the following differential problem:

$$\frac{dp_c}{dt} + \left[ \underbrace{\frac{\gamma p_c}{A_c (h_0 - \eta)}}_{\text{exhalation}}, \underbrace{\frac{\gamma p_c}{A_c (h_0 - \eta)} \left( \frac{p_{0c}}{p_c} \right)^{1/\gamma}}_{\text{inhalation}} \right] \frac{A_{pto} (p_c - p_{0c})}{K_1} = \frac{\gamma p_c}{h_0 - \eta} \frac{d\eta}{dt}, \quad \text{with } p_c(0) = p_{0c}, \eta(0) = 0, \quad (7)$$

where  $p_{0c}$  is the pressure in the chamber when  $\eta = 0$ , coincident with the atmospheric  
 pressure. The system in eq.(7) represents a non-linear differential problem that can  
 be numerically integrated upon defining the water level time function  $\eta(t)$  inside the  
 chamber. This problem is usually linearized in the hypothesis that  $|\eta| \ll h_0$  and that the  
 pressure chamber  $|p_c - p_{0c}| \ll |p_{0c}|$ , obtaining the following linear differential problem:

$$\frac{d\tilde{p}_c}{dt} + \frac{\gamma p_{0c}}{A_c h_0} \frac{A_{pto}}{K_1} \tilde{p}_c = \frac{\gamma p_{0c}}{h_0} \frac{d\eta}{dt}, \quad \text{with } \tilde{p}_c(0) = 0, \eta(0) = 0, \quad (8)$$

204 where  $\tilde{p}_c$  is the relative pressure in the chamber and  $p_{0c} = p_a$ .

205  
 206 If we indicate with the symbol  $r_{(\dots)}$  the ratio between the value of the variable  $(\dots)$   
 207 in the model and in the prototype, respectively, with the exception of the main length  
 208 scale indicated with  $\lambda$ , imposing the dynamic similarity is equivalent to satisfying the  
 209 following two equations:

$$\frac{r_{\tilde{p}_c}}{r_t} = \frac{r_{\gamma} r_{p_{0c}} r_{A_{pto}}}{r_{A_c} r_{h_0} r_{K_1}} r_{\tilde{p}_c} = \frac{r_{\gamma} r_{p_{0c}}}{r_{h_0}} \frac{r_{\eta}}{r_t}, \quad (9)$$

210 which refer the aerodynamic part of the OWC, to be added to the classical Froude  
 211 similarity conditions for the hydrodynamic component. It is worth recalling that the  
 212 invariance of the Froude number, in the model and in the prototype, requires that:

$$r_t = r_v = \lambda^{1/2}, \quad (10)$$

213 where  $\lambda$  is less than unity for smaller than prototype models. Equation (10) implies  
 214 the following scaling of some relevant variables: for the pressure it results  $r_p = r_{\rho_w \lambda}$ ,  
 215 where  $\rho_w$  is the density of water and with  $r_{\rho_w} \approx 1$  since water is also used in the model,  
 216 although fresh water instead of salt water for OWC in the sea; for the flowrate it results  
 217  $r_Q = \lambda^{5/2}$ ; for the acceleration it results  $r_a = 1$ . See [34] for the Froude scaling of other  
 218 variables.

219  
 220 In similarity analysis, in theory, the number of unknowns exceeds the number of  
 221 constraining equations, ensuring a sufficient number of degrees of freedom and, therefore,  
 222 ease in selecting scales starting with the geometric scale, which is the most relevant  
 223 constraint in physical modelling being, generally the length scale  $\lambda < 1$  selected according  
 224 to the laboratory facilities. In practice, other constraints arise for reasons of practicality  
 225 and cost. Among these constraints, a particularly important one arises from the fact  
 226 that the ambient pressure (outside the chamber) is the same in the model and in the  
 227 prototype, forcing the condition  $r_{p_{0c}} = 1$ . In addition, by scaling the cross-section area  
 228 of the chamber as  $r_{A_c} = \lambda^2$ , eqs.(9) reduce to

$$\frac{r_{\bar{p}_c}}{\lambda^{1/2}} = \frac{r_\gamma r_{A_{pto}}}{r_{h_0} \lambda^2 r_{K_1}} r_{\bar{p}_c} = \frac{r_\gamma}{r_{h_0}} \frac{r_\eta}{\lambda^{1/2}}, \quad (11)$$

or

$$\left\{ \begin{array}{l} r_{h_0} = \frac{r_{A_{pto}}}{\lambda^{3/2} r_{K_1}} r_\gamma, \\ r_\eta = \frac{r_{A_{pto}}}{\lambda^{3/2} r_{K_1}} r_{\bar{p}_c} = \frac{r_{h_0}}{r_\gamma} r_{\bar{p}_c}. \end{array} \right. \quad (12)$$

229 Scaling  $r_{A_{pto}} = \lambda^2$  and the PTO coefficient as  $r_{K_1} = \lambda^{1/2}$ , results in  $r_{h_0} = r_\gamma \approx 1$  and  
 230  $r_\eta = \lambda$ . The condition of an invariant height of the chamber, in the model and in the  
 231 prototype, as claimed by [65], is often replaced by the condition of  $r_{h_0} = \lambda$  with the model  
 232 chamber connected to an additional chamber with volume equal to  $\Delta V_{c,m} = (\lambda^2 - \lambda^3) V_{c,p}$   
 233 (the subscripts 'p' and 'm' refer to 'prototype' and 'model', respectively). This similarity  
 234 condition has been often adopted and successfully tested, see, e.g., [37]. Note that, on the  
 235 basis of eq.(9), the scale of relative pressures can be chosen at will, affecting only the scale  
 236 of  $\eta$  and  $K_1$ , but we do not forget that, in the present analysis, we are neglecting radiance  
 237 effects (see [8]), assuming that the forcing  $\eta(t)$  in the chamber is known a priori, a forcing  
 238 that is instead calculated on the basis of wave motion outside the chamber considering  
 239 scattering and radiation components of the potential flow describing the wave field.

240  
 241 An alternative approach is to adopt a different scaling for the coefficient  $A_{pto}/K_1$ . By  
 242 imposing  $r_{A_{pto}}/r_{K_1} \equiv r_{(A_{pto}/K_1)} = \lambda^{5/2}$ , with a Froude scaling for the pressure,  $r_{\bar{p}_c} = \lambda$ ,

243 results in

$$\begin{cases} r_{h_0} = \lambda r_\gamma, \\ r_\eta = \lambda^2, \end{cases} \quad (13)$$

244 which requires a chamber height in the model slightly smaller than  $\lambda h_{0,p}$  since  $r_\gamma \leq 1$ ,  
 245 and avoids the need for the additional volume, as previously pointed out by other authors  
 246 ([65, 12]). The vertical displacement of the water in the chamber of the model is reduced  
 247 with respect to the classical  $\lambda \eta_p$  value. Again, this approach is neglecting the interaction  
 248 between the water dynamics in the chamber and the external wave field, and can be  
 249 applied only for energetic sea state unless  $\lambda$  is quite large; for instance, by assuming  
 250  $\lambda = 1/10$  results  $r_\eta = 1/100$  and a very small amplitude of the water oscillation in the  
 251 chamber of the model equal to  $\eta_{0,m} = 0.5$  cm for  $H_p = 1$  m, which does not make sense.

252

### 253 3.2. The analysis for the full non-linear problem

Up to this point, we have investigated similarity conditions with reference to a linearized model. We now consider the similarity for the full non-linear process expressed by

$$\frac{d\tilde{p}_c}{dt} + \left[ \underbrace{\frac{\gamma(\tilde{p}_c + p_{0c})}{A_c(h_0 - \eta)}}_{\text{exhalation}}, \underbrace{\frac{\gamma(\tilde{p}_c + p_{0c})}{A_c(h_0 - \eta)} \left( \frac{p_{0c}}{\tilde{p}_c + p_{0c}} \right)^{1/\gamma}}_{\text{inhalation}} \right] \frac{A_{pto}\tilde{p}_c}{K_1} = \frac{\gamma(\tilde{p}_c + p_{0c})}{h_0 - \eta} \frac{d\eta}{dt}, \quad \text{with } \tilde{p}_c(0) = 0, \eta(0) = 0, \quad (14)$$

254 where a linear characteristic of the PTO is assumed.

255

256 For the exhalation process, we obtain the following similarity conditions:

$$\frac{r_{\tilde{p}_c}}{r_t} = \frac{r_\gamma r_{\tilde{p}_c}}{r_{A_c} r_{h_0}} \frac{r_{A_{pto}} r_{\tilde{p}_c}}{r_{K_1}} = \frac{r_\gamma r_{\tilde{p}_c}}{r_{h_0}} \frac{r_\eta}{r_t}, \quad \text{with } r_{h_0} = r_\eta, r_{p_{0c}} = r_{\tilde{p}_c}, \quad (15)$$

257 which cannot be satisfied since the condition  $r_{p_{0c}} = 1$  forces  $r_{\tilde{p}_c} = \lambda = 1$ , admitting only  
 258 the trivial solution  $\lambda = 1$ .

259

260 For the inhalation process, the same conditions (15) hold, plus the additional constraint  
 261  $r_\gamma = 1$  deriving from the additional contribution in the mass flowrate through the  
 262 PTO. Again, an exact similarity cannot be obtained.

263

264 If we consider a polynomial characteristic of the PTO:



$$\frac{d\tilde{p}_c}{dt} + \left[ \underbrace{\frac{\gamma(\tilde{p}_c + p_{0c})}{A_c(h_0 - \eta)}}_{\text{exhalation}}, \underbrace{\frac{\gamma(\tilde{p}_c + p_{0c})}{A_c(h_0 - \eta)} \left( \frac{p_{0c}}{\tilde{p}_c + p_{0c}} \right)^{1/\gamma}}_{\text{inhalation}} \right] \frac{\tilde{p}_c}{|\tilde{p}_c|} \frac{A_{pto} \left( \sqrt{K_1^2 + 4K_2|\tilde{p}_c|} - K_1 \right)}{2K_2} = \frac{\gamma(\tilde{p}_c + p_{0c})}{h_0 - \eta} \frac{d\eta}{dt}, \quad \text{with } \tilde{p}_c(0) = 0, \eta(0) = 0, \quad (16)$$

265 the similarity conditions are:

$$\frac{r_{\tilde{p}_c}}{r_t} = \frac{r_\gamma r_{\tilde{p}_c}}{r_{A_c} r_{h_0}} \frac{r_{A_{pto}} r_{K_1}}{r_{K_1}} = \frac{r_\gamma r_{\tilde{p}_c}}{r_{h_0}} \frac{r_\eta}{r_t}, \quad \text{with } r_{h_0} = r_\eta, r_{p_{0c}} = r_{\tilde{p}_c}, r_{K_1} = r_{K_2}^{1/2} r_{\tilde{p}_c}^{1/2}. \quad (17)$$

266 Again, only the trivial solution  $\lambda = 1$  is possible if  $r_{p_{0c}} = 1$ .

267

268 In conclusion, the linearized process allows scaling to compensate for the constraint  
 269  $r_{p_{0c}} = 1$  (i.e., the atmospheric pressure is the same, in the model and in the prototype);  
 270 the full non-linear model does not allow this correction and necessarily brings scaling  
 271 effects that are all the more relevant the more non-linearity is involved and the smaller  
 272 the geometric scale.

273

274 It is convenient to highlight concepts related to the assumption that the process is  
 275 adiabatic/isentropic. There is general agreement on the essentially adiabatic nature of  
 276 the air expansion–compression process, considering the wave cycle period is small enough  
 277 to prevent a complete heat exchange with the environment and boundaries —see Falcão  
 278 & Justino [16] as example—. This hypothesis helps to simplify the pressure coupling  
 279 through the continuity equation in the radiation–diffraction formulation and provides  
 280 with a clear approach to the air compression–expansion analysis. However, deviations  
 281 from strictly adiabatic conditions only, and further effects of moisture affecting the nature  
 282 of the gas inside the chamber, lead to a more accurate view when focusing on possible  
 283 causes for the low efficiency values observed in full scale prototypes. From the standpoint  
 284 of First and Second Principles of Thermodynamics, which underlie the formulation of the  
 285 energy–heat budget involved in compression–expansion, time as a variable is obviously  
 286 missing in the definition of state functions representing equilibrium states. As far as  
 287 the scale is concerned, it is clear that time scales involving both wave period and heat  
 288 exchange in order to reach thermal equilibrium (prescribed by the Zero Principle of  
 289 Thermodynamics) might lead to situations in which time required for heat exchange  
 290 could be balanced with wave period depending on the scale factor, hence biasing the  
 291 system performance from strictly adiabatic. A scale analysis reveals the extent to which  
 292 those effects, in turn associated with time, can be negligible.

#### 293 4. Other scale effects affecting the thermodynamics

294 Other issues related to the scale effects that affect OWC devices and related to the  
 295 thermodynamics process of the OWC will be exposed below.

296 *4.1. Similarity conditions for the water side*

297 If the study of the OWC relates only to gas dynamics, the flow field of the water is  
 298 assumed to be known within the chamber, and the water can be replaced, for example,  
 299 by a piston with an assigned law of motion; equivalently, the dynamics of the liquid  
 300 column in the chamber can be scaled almost arbitrarily. If, on the other hand (a rather  
 301 frequent situation) it is the overall behaviour of the OWC that is of interest, including  
 302 the interaction between air dynamics in the OWC and the wave field forcing the vertical  
 303 oscillation of the water interface in the chamber, then the similarity of the water phase  
 304 must also be considered. This similarity is Froude’s similarity, naturally arising due  
 305 to the fact that the restoring force of the water free surface is gravity, which faces the  
 306 convective inertia of water.

307  
 308 We briefly recall that under the assumptions of water-wave theory, a potential can be  
 309 used to describe the flow field, with  $\mathbf{v} = \nabla\phi$ , which satisfies the Laplace equation in the  
 310 domain,  $\nabla^2\phi = 0$ , the condition of impermeability at the rigid walls,  $(\partial\phi/\partial n = 0$  where  
 311  $n$  is the normal at the wall) and the condition that the free surface is a trajectory where  
 312 the Bernoulli theorem (neglecting the kinetic head) requires that

$$\eta - \frac{1}{g} \frac{\partial\phi}{\partial t} \Big|_{fs} = \begin{cases} \frac{\tilde{p}_c}{\rho_w g}, & \text{in the chamber,} \\ 0, & \text{out of the chamber.} \end{cases} \quad (18)$$

313 The next steps are based on Evans’ method, which consists of decomposing the  
 314 potential into the sum of a scattering component and a radiation component —see  
 315 reference [10]—. Subsequent analysis would lead to calculating the two potentials, thus  
 316 finding the solution to the problem that couples the dynamics of the air column in the  
 317 chamber to the dynamics of water in the fluid domain. In practice, the wave field is  
 318 distorted by the presence of the OWC. As a result of this, the fluctuation of the water  
 319 column in the chamber is not known a priori, but is a non-linear function of the coupling  
 320 between water column and air column.

321  
 322 Eq.(18) can be rearranged by decoupling the two variables  $\eta$  and  $\phi$ , making use of  
 323 the kinematic condition at the free surface:

$$\frac{\partial\eta}{\partial t} - \frac{\partial\phi}{\partial z} \Big|_{fs} = 0, \quad (19)$$

324 obtaining

$$\frac{\partial\phi}{\partial z} \Big|_{fs} - \frac{1}{g} \frac{\partial^2\phi}{\partial t^2} \Big|_{fs} = \begin{cases} \frac{1}{\rho_w g} \frac{\partial\tilde{p}_c}{\partial t}, & \text{in the chamber,} \\ 0, & \text{out of the chamber.} \end{cases} \quad (20)$$

325 The similarity conditions for the process described at the free surface by eq.(20) are

$$\left\{ \begin{array}{l} \frac{r_\phi}{\lambda} = \frac{r_\phi}{r_t^2} = \frac{r_{\bar{p}_c}}{r_{\rho_w} r_t}, \quad \text{in the chamber,} \\ \frac{r_\phi}{\lambda} = \frac{r_\phi}{r_t^2}, \quad \text{out of the chamber,} \end{array} \right. \quad (21)$$

326 where by definition of potential results  $r_\phi = r_v \lambda$ . The similarity conditions result in

$$r_t = \lambda^{1/2}, r_{\bar{p}_c} = r_v \lambda^{1/2} r_{\rho_w}, \quad (22)$$

327 which permanently link the pressure scale in the chamber to the velocity scale in the  
 328 water column.

#### 329 4.2. Similarity for the turbine

330 In the experimental approach for studying OWCs, it is common to replace the turbine  
 331 with a hole or porous septum, which determine a quadratic ( $\Delta p \propto v^2$ ) or linear ( $\Delta p \propto v$ )  
 332 characteristic to simulate different types of turbines commonly in use. It is obvious that  
 333 the characteristics of real-world turbines have a more complex functional structure, with  
 334 torque, efficiency and resistance curves, which require bench measurements. A more  
 335 complete analysis also requires the modelling of the generator, and is ultimately framed  
 336 in a wave-to-wire model. In particular, the behaviour of the turbines also depends on  
 337 the generator and the control system, in a model in which a large number of variables  
 338 intervene that depend on both the turbine model adopted and the control system.  
 339 Consider, in this respect, what is detailed in [26], in an analysis in which the numerous  
 340 aspects that condition the overall efficiency of the system are analysed, including the  
 341 behaviour of the turbines and the generator.

342  
 343 In a detailed analysis of the OWC, it is also imperative to adequately reproduce the  
 344 turbine dynamics, which are characterized by sometimes very small scaling ratios. For  
 345 example, aerodynamic forces and inertial forces, expressed as:

$$F_{aer} = \frac{1}{2} \rho_a C_r v^2 A^2, \quad F_{in} = \rho_a V \ddot{x}, \quad (23)$$

346 are scaled as

$$r_{F_{aer}} = r_v^2 \lambda^2, \quad r_{F_{in}} = \lambda^3 r_v r_t^{-1} \rightarrow r_{F_{aer}} = r_{F_{in}} = \lambda^3 \quad (24)$$

347 in Froude similarity and assuming that the  $C_r$  has the same value in the model and in  
 348 the prototype. With similar reasoning, the inertia of the rotor scales as  $r_I = \lambda^5$ , the mass  
 349 of the blade scales as  $r_m = \lambda^3$ , the power scales as  $r_P = \lambda^{7/2}$ , the torque scales as  $r_T = \lambda^4$ .

350  
 351 However, in practical applications, it is difficult to construct the turbine in such a way  
 352 that it respects scaling, since the mass of the propeller, for example, is usually too small  
 353 and the inertia of the rotor is also difficult to scale correctly, unless  $\lambda$  is not very small. In  
 354 one of the few tests in literature carried out using a geometrically scaled impulse turbine  
 355 with speed control through a servo-motor ([33]), the dimensions of the impulse turbine  
 356 model could not be reduced to match the optimum damping ratio of the orifice. Turbine

357 speed control is equivalent to an orifice with a variable diameter: as the rotation speed  
358 increases, the pressure drop also increases. In addition, the oversized turbine model  
359 also results in efficiencies that cannot be optimized in the OWC laboratory models, but  
360 which can be used to validate advanced numerical models of the chamber-turbine system  
361 and wave-to-wire models. Finally, the efficiency of the propeller blades is different in  
362 the model and in the prototype since the Reynolds number of the air is smaller in the  
363 model than in the prototype. To obviate, for example, the lower efficiency blade, it may  
364 be appropriate to change the shape of the profile, taking a tip from the vast literature  
365 originating from the development of drone blades, which are evidently characterised by  
366 low Reynolds operation.

367  
368 An insurmountable scaling effect arises from friction, which is notoriously non-scalable  
369 and ends up playing a dominant role the smaller  $\lambda$  is. This means that regardless of  
370 the construction materials adopted for the turbine, the adjustments that can be used  
371 to make airfoils that, at lower Reynolds numbers, have the same efficiency as the real  
372 airfoils, friction remains as a disturbing cause, reducing efficiency in the model much more  
373 than in the prototype. It is conceivable that a servo-driven turbine, with a controlled  
374 motor capable of reproducing the transient dynamics of real turbines to scale, could be  
375 a solution in cases where the complete simulation of the turbine becomes important for  
376 the model study of the OWC. See, e.g., [30] for an application of a hardware-in-the-loop  
377 approach to control wind turbines.

378  
379 Attempting to implement all the information on the basic formulation might lead  
380 to conclusions that, in turn, can be hiding some relevant features. We have focused on  
381 the scale effect affecting the thermodynamic problem, as a feasible way to overcome the  
382 fact that, strictly speaking, time is not a variable included in the formulation of state  
383 variables, First and Second Principles and heat and energy budgets. Focusing on the  
384 scale effects, which in turn are inherently affected by scaled time, if not a complete way  
385 to implement time in the Thermodynamics formulation, it is feasible way to look into  
386 what relative differences might be expected when dealing with different scale prototypes.

387  
388 On the other hand, it is important to highlight the difficulty of the construction  
389 of a scaled turbine connected to an electricity generator. Most of the experimental  
390 research have replaced the turbine with a porous septum or an orifice —see Thibeaut  
391 *et al.* [62], López *et al.* [37], Sheng *et al.* [59], Bingham *et al.* [1]— Even the Authors of  
392 the present research conducted numerical research using an actuator disk model —see  
393 Medina-Lopez *et al.* [40]—. Nevertheless, as a first approach, Authors of this research  
394 have performed some experimental test using a turbine, which implies to modify the  
395 relationship between pressure drop and air flow through it (linear in the case of the  
396 turbine, quadratic in the case of the porous septum). The next step would be to connect  
397 the turbine to an electric generation system, but this is not an easy task due to the  
398 friction induced to the turbine by the generator system, which can easily lead to an  
399 out-of-scale turbine model performance.

400  
401 In addition, according to previous research by the authors —Molina *et al.* [46]—, the  
402 turbine acts like a restraint to the thermodynamic system. Therefore, its characteristics  
403 affect the thermodynamic compression–expansion process and, consequently, affect to

404 the overall process. So, the replacement of the turbine with an orifice or porous septum  
405 would affect not only to the scale effects of the system, but to the overall performance  
406 of the device.

407

408 In conclusion, while a detailed analysis of the individual components is permissible  
409 as a first step, only an analysis of the entire OWC system allows the interdependencies  
410 between them to be studied. Suffice it to say that the damping of the OWC structure  
411 depends on both the geometry and the operating point of the turbine, which in turn  
412 is defined by a strategy to optimize the overall efficiency and power: every detail is  
413 important in order to determine with sufficient accuracy the efficiency of the entire  
414 system.

415

#### 416 *4.3. Scaling of turbulence and the effects of Reynolds, Weber and Mach numbers*

417 One aspect of scaling that is practically always overlooked is turbulence. The classical  
418 study of turbulence identifies a series of geometric and temporal scales, which are coupled  
419 by defining velocity scales. In this sense, the book by [61] is a clear example of a physical  
420 interpretation of turbulence on the basis of scales.

421

422 Turbulence in an OWC plays a major role and varies during the two exhalation–  
423 inhalation phases: in the first, turbulence is generated by the sloshing process and is  
424 strongly modulated in the compression phase, during which the vortices interact in a  
425 forced manner presumably different from the classical cascade scheme; in the second,  
426 atmospheric turbulence, near the turbine inlet, modulates the conveyed flow and invades  
427 the chamber after interaction with the blades. In both cases, at prototype scale, tur-  
428 bulence is seldom homogeneous and isotropic, and the spectrum deviates significantly  
429 from the classical Kolgomorov equilibrium spectrum. This also happens in the model,  
430 but it is intuitive that the scaling of variables is anything but straightforward and simple.

431

432 What is most interesting about the phenomena in the OWC, is the turbulent mixing  
433 that is coupled to the dynamics, defined as Level 2 in [9]. The most relevant aspect  
434 of that phenomena is the generation of baroclinic vorticity, due to the misalignment  
435 between pressure gradient and density gradient, i.e. between temperature gradient and  
436 entropy gradient. Vorticity of this nature facilitates the development of Kelvin-Helmoltz  
437 layers and consequent instability, with major effects on mixing. In this case, the coupling  
438 between mixing and flow field dynamics is due to the mixing’s ability to reduce gradients,  
439 altering, in feedback, the generation of vorticity. If we want to evaluate these effects on  
440 scaling an OWC, it is intuitive that, for the same fluid (air, in the case of an OWC), the  
441 geometric size of the chamber is relevant in determining the level of heterogeneity, which  
442 is quite different for a full-scale OWC than for a reduced geometric scale model OWC.  
443 This means that some mixing mechanisms are not reproduced homothetically, leading to  
444 different process scales between prototype and model.

445

446 In practice, the structure of the turbulence is strongly influenced by the Reynolds  
447 number, which, in Froude similarity, scales according to  $r_{Re} = \lambda^{3/2}$ , being smaller in  
448 the model than in the prototype if  $\lambda < 1$  and if  $r_\nu = 1$ , where  $\nu$  is the kinematic  
449 viscosity. This applies to both the air and water side, the former being more important

450 for the thermodynamic evolution of the system. Reducing the Reynolds number results  
451 in smaller time scales in the model than in the prototype if  $r_{Re} = 1$ . It also entails  
452 proportionally larger geometric scales: the separation of micro-vortices from macro-  
453 vortices is sharper if the Reynolds number is high. Since macro-vortices contain most  
454 of the energy, and micro-vortices contain most of the vorticity, if, in the transition from  
455 prototype to model, the ratio of density between the two classes of vortices varies, the  
456 distribution of energy and vorticity as a function of frequency (or rather, of the wave  
457 number) also varies accordingly.

458  
459 The consequences of the scale effect on the distribution of energy and vorticity  
460 are quite relevant if we consider the transport processes (of heat, momentum, etc.)  
461 especially in the gas phase, which is the most thermodynamically active during cycles  
462 of an OWC. From this point of view, the diffusion of heat generated by the process of  
463 compressing the air in the chamber is commonly schematised by the Boussinesq model,  
464 assuming that it is proportional to temperature gradients through the thermal diffusivity,  
465 a phenomenological parameter similar to turbulent diffusivity. If the Reynolds number in  
466 the model is smaller than the Reynolds number in the prototype, the spatial gradients of  
467 the variables such as temperature, velocity, etc., will be smaller than they should be (this  
468 pattern is visually consistent with a more 'coarse' structure of turbulence at low Reynolds  
469 numbers) and thus the heat fluxes in the model will be smaller than they should be. We  
470 also remind that the Reynolds number can also be interpreted as the ratio of turbulent  
471 diffusivity to molecular diffusivity,  $Re = uL/\nu \equiv \nu_T/\nu$  and thus smaller Reynolds in  
472 the model than in the prototype inevitably reduce the speed of momentum (and other  
473 variables) diffusion. It holds also for heat, entropy, and all the other quantities involved  
474 in the transformation.

475  
476 During the inhaling process, the situation is even more complex. In the prototype,  
477 atmospheric turbulence is often quite intense, especially in the more energetic sea states  
478 usually accompanied by wind storms and bursts of turbulence. Depending on the mea-  
479 sures taken to shield the turbine outlet, the flow entering in the chamber has a more or  
480 less high turbulence level, unlike in the model, which is normally tested in the absence  
481 of wind (unless a wind-wave tunnel is used) and therefore with zero or very small initial  
482 turbulence level. In addition, the turbulent flow of air at atmospheric pressure invades  
483 the chamber which, in the prototype, is full of air in depression characterised in any case  
484 by a non-negligible level of turbulence, while in the model it has a correspondingly lower  
485 level of turbulence than it should. The effect on turbulence due to the propeller blades is  
486 also present in the prototype, where the blades induce swirling (unless counter-rotating  
487 double propeller turbines are used) and, anyway, distorce turbulence, while the turbine  
488 is rarely installed in the model, due to the difficult scaling of certain variables such as  
489 rotor inertia and blade Reynolds number. To give an idea of the difficulties in stating  
490 similarity rules for similar cases, a summary of the complex scalings required for air  
491 turbulence and water turbulence during wind wave generation can be found in [6].

492  
493 Another aspect to consider is the scaling of the Weber number, which is clearly not  
494 unitary if water is also used in the model since, in Froude similarity it results  $r_{We} = \lambda^2$ .  
495 A relatively low surface tension facilitates the incorporation of air into the water phase  
496 and the generation of droplets in the air phase. In the model, on the other hand, the size

497 of the eddies and the turbulent velocity scale are too small for dominating the surface  
498 tension and therefore both foam and droplets are not or are rarely present. This has  
499 consequences, in the OWC chamber, mainly for the thermodynamics of the gas, since the  
500 gas lacks the characteristic spray and therefore has different thermodynamic properties  
501 than in the real world.

502  
503 In addition, we remind that the Mach number also scales according to  $r_{Ma} = \lambda^{1/2}$  in  
504 Froude similarity, being smaller in the model than in the prototype. This means that,  
505 in addition to the compressibility of air, which we have already discussed at length, the  
506 pressure waves that inevitably characterise water and air in the chamber also have a more  
507 damped effect in the model than in the prototype. The shock phenomena that might  
508 occur in the sloshing of air-water mixture in the chamber of the real OWC device (with  
509 consequent dissipation of energy) ([52]) certainly do not occur in the model, introducing  
510 an additional scaling effect.

511  
512 All in all, while it is difficult to quantify the scale effects on turbulence and it is  
513 impracticable to eliminate them (it would require a gas, in the model, with kinematic  
514 viscosity reduced by a factor of  $\lambda^{3/2}$  compared to air), it is immediately apparent  
515 that the thermodynamic transformations that occur, and which in physical reality are  
516 always non-equilibrium, are also more so in the prototype than in the model. As a  
517 consequence, classical thermodynamics based on quasi-equilibrium states works with a  
518 different approximation level for the model than for the prototype, and a non-equilibrium  
519 thermodynamics approach is more suitable (see, e.g., [32]). Non-equilibrium thermody-  
520 namics demands to be implemented for a number of good reasons, i) it provides an  
521 accurate description of the coupled transport processes; in the case of OWC we have  
522 already classified the quantities transported, i.e. mass, heat, moisture; ii) it quantifies  
523 the production of entropy, lost work and lost exergy; iii) it provides the entropy budget  
524 to be used in thermodynamic modelling. These conclusions should be taken into account  
525 when extrapolating laboratory data to the real data.

526  
527 The foregoing discussion reveals the influence of the scale factor in the dimensional  
528 variables governing the problem. However, for thermodynamic system parameters the  
529 eventual influence of scale, i. e. the reference volume size of the gas system enclosed in  
530 the chamber, might not be so evident. The scale effect can modify the thermodynamic  
531 response through parameters that are not explicitly dependent on the system volume,  
532 hence on the representative length scale. That is the case of the polytropic exponent  
533 defining the system process equation —see equation (4)—. While a first approach might  
534 lead to assume  $r_\gamma = 1$  following the non dimensional nature of the polytropic, it will be  
535 shown later in §5.1 that a dependence on the length scale can be formulated through  
536 non-equilibrium instability approach.

537  
538 In conclusion, all these scale effects, which cannot be eliminated for reasons of cost or  
539 because there are no fluids matching with the scale requirements, such as density, viscos-  
540 ity, surface tension or compressibility, must nevertheless make extrapolations of model  
541 measurements to the real thing extremely cautious, and push towards the realization of  
542 models at scales that are not excessively small. The classic suggestion to make at least  
543 two models with different geometric scales, so as to estimate the trend of the scale effects

544 in order to extrapolate the correct results to reality, is still valid and appropriate, even  
 545 if it comes up against a doubling of the experimental workload and costs.

## 546 5. Instability analysis

547 The air expansion–compression process in the OWC device follows a polytropic pro-  
 548 cess characterized by the polytropic exponent  $n$  as indicated in equation (4). In this  
 549 case, if the thermodynamics is to be affected by any scale effects to be considered in the  
 550 experimental tests, those effects might be represented through the polytropic exponent  
 551  $n$  and its intrinsic dependence on system variables, that in turn can be affected by the  
 552 scale of the problem, e. g. the system volume.

553 Indeed, Thermodynamics essentially deals with equilibrium states, with state func-  
 554 tions not defined for transient ones. However, many process could never reach an  
 555 equilibrium state in a strict sense, neither because the nature of the process itself, nor  
 556 because the size of the system. For example, let us consider a process where there is a  
 557 cyclical heat exchange. Therefore it is reasonable to think that the smaller the system  
 558 the faster the thermal equilibrium can be reached. In the case of the air expansion–  
 559 compression process of the OWC device, some scale effects could appear during scale  
 560 model tests. Hence, an instability analysis applied to the polytropic exponent around an  
 561 equilibrium volume  $V_0$ , can reveal some information on such effects.

562 The most general expression of the polytropic exponent is:

$$n = \frac{m}{K_T p} \quad (25)$$

565 where  $m$  is the polytropic index —which is a relation between the specific heat under  
 566 constant pressure, volume and a certain variable  $y$ —,  $K_T$  is the isothermal compressibility  
 567 coefficient, and  $p$  is the pressure. The specific heat under any variable  $x$  is defines as  
 568  $C_x = T(\partial S/\partial T)_x$ , being  $S$  the entropy and  $T$  the temperature. So, the polytropic  
 569 index depends on these two variables,  $m = f(S, T)$ . On other hand, the isothermal  
 570 compressibility coefficient is a function of the volume and the pressure,  $K_T = f(p, V)$ .  
 571 So, the polytropic exponent is  $n = f(S, T, V, p)$ . Nevertheless, if the dependence of  $n$   
 572 with the volume is taken into account, it is being assumed that there is a scale effect.  
 573 So, the volume dependence will not be taken into account. The pressure dependence is  
 574 cancelled with  $K_T$ , so finally,  $n = f(S, T)$ .

575 The Taylor series of the polytropic exponent around a initial volume  $V_0$  can be  
 expressed as:

$$n(V) = n(V_0) + \left. \frac{\partial n}{\partial V} \right|_{V_0} (V - V_0) + \frac{1}{2} \left. \frac{\partial^2 n}{\partial V^2} \right|_{V_0} (V - V_0)^2 + \dots \quad (26)$$

576 The polytropic exponent is a function of the entropy and the temperature, so:

$$\frac{\partial n}{\partial V} = \left( \frac{\partial n}{\partial S} \right)_T \left( \frac{\partial S}{\partial V} \right)_T + \left( \frac{\partial n}{\partial T} \right)_S \left( \frac{\partial T}{\partial V} \right)_S \quad (27)$$



577  
578

On the other hand,  $n = f(m, K_T, p)$ , so:

$$\begin{aligned} \left(\frac{\partial n}{\partial S}\right)_T &= \left(\frac{\partial n}{\partial m}\right)_T \left(\frac{\partial m}{\partial S}\right)_T + \left(\frac{\partial n}{\partial K_T}\right)_T \left(\frac{\partial K_T}{\partial S}\right)_T + \left(\frac{\partial n}{\partial p}\right)_T \left(\frac{\partial p}{\partial S}\right)_T \\ \left(\frac{\partial n}{\partial T}\right)_S &= \left(\frac{\partial n}{\partial m}\right)_S \left(\frac{\partial m}{\partial T}\right)_S + \left(\frac{\partial n}{\partial K_T}\right)_S \left(\frac{\partial K_T}{\partial T}\right)_S + \left(\frac{\partial n}{\partial p}\right)_S \left(\frac{\partial p}{\partial T}\right)_S \end{aligned}$$

579  
580

Substituting these expression into (27), the first term of the Taylor expansion is:

$$\frac{\partial n}{\partial V} = \frac{1}{K_T p} \left[ \left(\frac{\partial m}{\partial V}\right)_S + \left(\frac{\partial m}{\partial V}\right)_T \right] - \frac{m}{K_T^2 p} \left[ \left(\frac{\partial K_T}{\partial V}\right)_S + \left(\frac{\partial K_T}{\partial V}\right)_T \right] - \frac{m}{K_T p^2} \left[ \left(\frac{\partial p}{\partial V}\right)_S + \left(\frac{\partial p}{\partial V}\right)_T \right] \quad (28)$$

581  
582  
583

Naming as  $n' = \partial n / \partial V$  in order to simplify, the second term of the Taylor expansion is:

$$\begin{aligned} \frac{\partial^2 n}{\partial V^2} &= \frac{\partial n'}{\partial V} = \left(\frac{\partial n'}{\partial S}\right)_T \left(\frac{\partial S}{\partial V}\right)_T + \left(\frac{\partial n'}{\partial T}\right)_S \left(\frac{\partial T}{\partial V}\right)_S = \left(\frac{\partial n'}{\partial m}\right) \left[ \left(\frac{\partial m}{\partial V}\right)_S + \left(\frac{\partial m}{\partial V}\right)_T \right] + \\ &+ \left(\frac{\partial n'}{\partial K_T}\right) \left[ \left(\frac{\partial K_T}{\partial V}\right)_S + \left(\frac{\partial K_T}{\partial V}\right)_T \right] + \left(\frac{\partial n'}{\partial p}\right) \left[ \left(\frac{\partial p}{\partial V}\right)_S + \left(\frac{\partial p}{\partial V}\right)_T \right] \end{aligned} \quad (29)$$

#### 584 5.1. Ideal gas, adiabatic process

585 To get a specific expression of the Taylor expansion from the general expression (26)  
586 and from the computed terms (28) and (29), the type of process must be known. In a  
587 first approach, let us consider an adiabatic and reversible process of an ideal gas, which  
588 state equation is  $pv = R_0 T$ , where  $v$  is the molar volume. In this case, the entropy is  
589 constant, so  $C_y = C_s = 0$ ,  $K_T = 1/p$  and  $n = m = 1.4$ . Applying the state equation to  
590 the Taylor expansion, taking into account that the derivative of  $K_T$  with the volume is  
591 null and  $m$  is constant, the first and second term of this expansion is:

$$\begin{aligned} \frac{\partial n}{\partial V} &= \frac{-m}{K_T p^2} \frac{-2p}{V} = \frac{2m}{K_T p V} = \frac{2n}{V} \\ \frac{\partial^2 n}{\partial V^2} &= \frac{-2m}{K_T p^2 V} \frac{-2p}{V} = \frac{4m}{K_T p V^2} = \frac{4n}{V^2} \end{aligned}$$

592 Finally, the Taylor expansion of the polytropic exponent for the adiabatic process is  
593 cleared out in equation (30):

$$n(V) = n(V_0) + \frac{2n(V_0)}{V_0} (V - V_0) + \frac{2n(V_0)}{V_0^2} (V - V_0)^2 \quad (30)$$

594 Equation (30) represents a system with reference volume  $V_0$  which is essentially  
595 governed by a reference polytropic exponent  $n(V_0)$ . From a thermodynamic point of view,

596  $n$  should not be expected to change according with its non-dimensional nature. However,  
 597 the Taylor expansion allows to set a dependence between the polytropic exponent and  
 598 the system volume variations, which can be used to approach the influence of the system  
 599 volume scale on the thermodynamic performance. This dependence is represented in  
 600 Figure 1, where the variation of the polytropic exponent is represented against the air  
 601 volume in non-dimensional form.

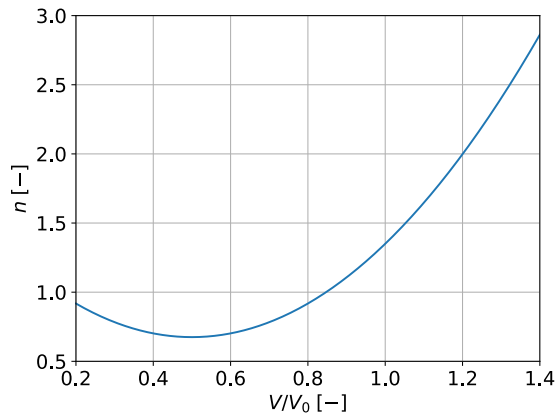


Figure 1: Variation of the polytropic exponent with the non-dimensional volume of the system, according to the instability analysis indicated in eq.(30).

602 It is clear that according to the rationale, any change with respect to the reference  
 603 volume  $V_0$ , i. e.  $V/V_0 \neq 1$ , is associated with a change in  $n$ . It can be observed how  
 604 the system becomes more sensible to the volume variation in terms of the polytropic  
 605 exponent for larger values of  $V/V_0$ , say for the greater values of system volume. This  
 606 different sensibility could be interpreted as a variation in the thermodynamic processes  
 607 with the variation in scale, affecting the OWC device performance and its efficiency.

608  
 609 It is clear that the nature of the dependence of the polytropic exponent with the  
 610 system volume variation as represented in equation (30), is fixed by the form of the  
 611 Taylor expansion. However, on the ground of that dependence, it can be deduced a  
 612 thermodynamic performance which helps to explain how an increase in volume can affect  
 613 the nature of the compression/expansion process through the polytropic equation (4)  
 614 governing it. From a purely qualitative point of view, expression (30) and Figure 1 reveal  
 615 that as the ratio  $V/V_0$  increases, the variation of the polytropic exponent becomes more  
 616 noticeable. The previous statement is in turn coherent with the fact that non-equilibrium  
 617 states are intrinsically related with the time-length dimensions involved in the even  
 618 distribution of state function values over the system volume. Indeed, that conclusion  
 619 reveals that for an OWC model to be representative of the full-scale thermodynamics, it  
 620 would require a different enlarged scale for the system volume dimension, so that transient  
 621 stated in-between equilibrium states to be expected at full-scale, be represented in a

622 more realistic way. This point has been previously suggested by some authors, [65, 12].  
 623 In any case, whether the deviation of the polytropic exponent from equilibrium values  
 624 entirely addresses the theoretical approach in figure 1, or requires further enhancement  
 625 implementing additional factors, is an open line by the authors of this research.

### 626 5.2. Case study

627 Let us consider a full scale OWC device with a system air volume  $V_0$ . Following  
 628 the discussion in §5.1, the variation of the polytropic exponent due to the air volume  
 629 variations can be estimated. It can be seen from Figure 1 that a shift in this curve can  
 630 mean either a change in the initial volume  $V_0$  due to a change in the escale, or a change  
 631 in the volume range for a given  $V_0$ . All in all, any volume oscillation around  $V/V_0 \neq 1$   
 632 can be interpreted so that the model has a different scale than the prototype.

633 Now a full-scale device with initial air volume  $V_0 = 30 \text{ m}^3$  is compared with a  
 634 scaled model with air volume  $V_{0,m} = V_0/2 = 15 \text{ m}^3$ . In both cases, a 20% air volume  
 635 variation around the initial air volume is applied, which can be a representation of the air  
 636 volume variation induced by waves. The full-scale device would shift between the values  
 637  $V/V_0 = [0.8, 1.2]$ , and the scaled model between the values  $V/V_0 = [0.4, 0.6]$ . That means  
 638 a polytropic exponent variation ranging between  $[0.91, 1.99]$  for the full-scale device, and  
 639  $[0.67, 0.70]$  for scaled model, as figure 2 shows.  
 641

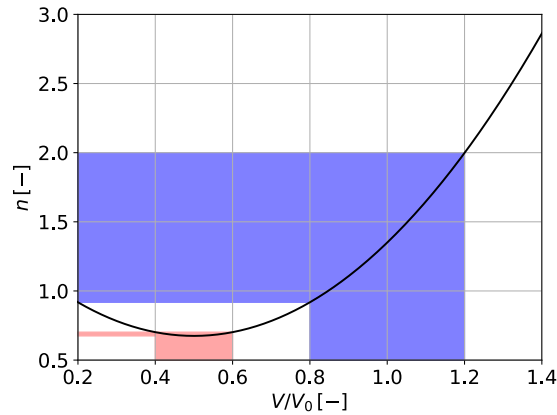


Figure 2: Variation of the polytropic exponent with the non-dimensional volume of the system (according to eq.(30)). The shaded areas indicates the two cases of study. The blue one represent the full-scale device, and the red one the scaled model.

642 Applying the polytropic process expression in the form  $pV^n = \text{const}$ , the pressure  
 643 variation for both cases can be estimated. The pressure values obtained are represented  
 644 in Figure 3. Figure 3(a) shows the variation of the pressure with the volume, following the  
 645 polytropic process. It can be observed that the range of pressure variation is different,

646 depending on the scale considered. The pressure variation range is wider for the full  
647 scale model than for the reduced scale model, as it was expected due to the range of  
648 variation of the polytropic exponent. Nevertheless, according to Froude similarity the  
649 relation between the pressures in the prototype and in the model should be linear, see  
650 section § 3, but Figure 3(b) shows that this dependence is not linear, which might reveal  
651 the existence of scale effects.  
652

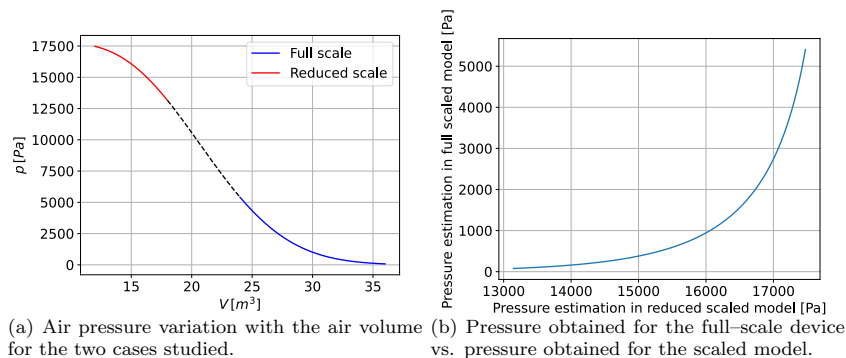


Figure 3: Pressure obtained for the two cases studied.

653 Those values of pressure can be used to estimate the efficiency of both devices. Let  
654 us consider a vertical cylindrical OWC chamber with 2.5 m diameter, which would mean  
655 an emergence height of 6.1 m for the prototype and 3.05 m for the scaled model. Now  
656 it is considered the implementation of a Wells turbine with performance characteristics  
657 similar to Pico plant, [18], with 2.3 m of diameter and a rotational speed of 1500 r.p.m.,  
658 whose calibration curve and efficiency curve are known. So, with the pressure, volume  
659 and polytropic exponent values estimated, the efficiency of the device for the two cases  
660 studied can be estimated using the mentioned calibration and efficiency curves. The  
661 estimated efficiency is 0.621 for the prototype, and 0.576 for the scaled model. If there  
662 were no scale effects, the efficiency in both cases should be the same —since the efficiency  
663 is a non-dimensional parameter—, so the differences in the efficiency for both cases can  
664 be due to the existence of scale effects, as it has been stated before. In fact, this result  
665 agrees with the proposal of [65] and [12] regarding the requirement for an increase in  
666 the scaled device volume of the air chamber. In any case, further research is required to  
667 implement thermodynamic effects whose scale dependence and extent is not trivial.  
668

669 In the case of a very small scale, such as the experimental test that can be performed  
670 in the laboratory, the results can be different, as all the evidence indicates. Following the  
671 same reasoning as before, if the initial volume were  $3 m^3$  ( $V_{0,s} = V_0/10$ ), the efficiency of  
672 the device would be  $\eta_s = 0.707$ . The values estimated in the previous paragraph indicate  
673 that  $\eta_p > \eta_m$ , so one would expect that the smaller the scale, the lower the efficiency  
674 obtained. However, the efficiency of the very small device is higher than that of the model

675 and prototype. Thus, scale effects may play an important role at very small scales. In  
676 addition, it is important to note that non-equilibrium states become more relevant as  
677 the scale increases. Thus, in the very small scale, these non-equilibrium states would not  
678 be as obvious.

679  
680 Similar results are obtained considering an adiabatic process for a real gas, whose state  
681 equation is  $pv = ZR_0T$ . To check the results for a real gas and a non-adiabatic process,  
682 the state equation must be known, as well as the variable  $y$  that remains constant, in  
683 order to obtain a specific expression of the Taylor series.

## 684 **6. Discussion, conclusions and future research**

685 In this research, a theoretical approach to a comprehensive understanding the scale  
686 effects in OWC devices has been made. The main advantages and disadvantages of this  
687 similarity analysis are:

- 688 • Regarding the dynamic and thermodynamic of the OWC, the linearized expansion-  
689 compression process allows scaling to compensate for the constraint that the atmo-  
690 spheric pressure be the same in the model and in the prototype.
- 691 • Meanwhile, the non-linear model does not allow this correction, which necessarily  
692 brings scaling effects that are all the more relevant the more non-linearity is  
693 involved and the smaller geometric scale.
- 694 • The similarity analysis brings to front the problems that appears when trying to  
695 couple the hydrodynamics, thermodynamics and aerodynamics process that occurs  
696 in the OWC scaled devices.
- 697 • In practical applications, it is difficult to construct a turbine that respects all the  
698 scale relations, like the aerodynamic and inertial forces, the inertia of the rotor,  
699 the mass of the blades, or the torque, among others.
- 700 • According to Froude similarity, the Reynolds number is smaller in the model than  
701 in prototype, which influences strongly the structure of the turbulences. This is  
702 quite relevant while considering the transport process (heat, momentum, etc.) and  
703 affecting strongly the thermodynamics process.
- 704 • The Mach number in the scaled model is smaller than in the prototype, according  
705 to Froude similarity. That means that the pressure waves have a more damped  
706 effect in the model than in the prototype.
- 707 • The scaling of the Weber number, which is not unitary following the Froude  
708 similarity, affects the thermodynamics of the gas inside the OWC chamber since  
709 the gas lacks the characteristics spray and therefore has different properties than  
710 in the prototype.
- 711 • The research brings to front the fact that, in the case of thermodynamic processes,  
712 the phenomena accuracy increases downward scaling from prototype to model, as  
713 oposite to other area of similitude analysis, in which accuracy increases upward

714 scaling from model to prototype. In that sense, the fine tuning of the air chamber  
715 scale, regardless the scale adjustment of the rest of variables involved on the  
716 problem, helps to increase the accuracy in the real full-scale phenomena from  
717 prototype to model.

718 Whether the scale effect is relevant when approaching a complete wave-to-wire is yet  
719 to be analyzed in depth. Not so much from a pure theoretical or numerical way as from  
720 an experimental set-up. In any case, this theoretical approach is intended to settle a refer-  
721 ence frame for future research on the topic. In addition, the results are consistent with  
722 previous experimental/theoretical studies, leading to a better understanding of the point  
723 that in the case of the thermodynamics process involved in air compression-expansion  
724 inside the chamber, scaled devices provide with a better framework for thermodynamic  
725 process to match equilibrium conditions, otherwise mandatory for the application of First  
726 and Second Principles of thermodynamics is obviously a counter effect to other processes  
727 involved in OWC performance, specially those related with the wave impingement,  
728 radiation-diffraction and turbulence, in which large-scale devices provide with more  
729 realistic representation of the phenomena. Even if experimental and numerical research  
730 are a feasible way to observe all of the above, a theoretical basis is required to set the  
731 guidelines.

732  
733 This research intends to set the basis for the next numerical simulations and exper-  
734 imental test. That new research is meant to focus on the study of the chamber and  
735 turbine size, where different configurations will be compared with the aim to check the  
736 scale effects. The main conclusions of this research are:

- 737 • In the PTO similarity, the linearized process allows scaling to compensate for the  
738 constraint that the atmospheric pressure be the same in the model and in the  
739 prototype. The full non-linear model does not allow this correction and necessarily  
740 brings scaling effects that are all the more relevant the more non-linearity is  
741 involved.
- 742 • The similarity conditions for the water side implies that the pressure scale in the  
743 chamber is linked permanently to the velocity scale in the water column.
- 744 • As for the similarity for the turbine, it is difficult to construct a turbine that  
745 respect simultaneously all the factors that affect its performance, like the inertia  
746 of the rotor, the mass of the propeller, or the friction effects, among others.
- 747 • The turbulence plays an important role that must be taken into account, and varies  
748 during the two exhalation/inhalation phases. While it is difficult to quantify the  
749 scale effects on turbulence, these effects affect to the thermodynamics process.
- 750 • A comprehensive approach through instability analysis reveals that the scale size  
751 introduces differences in the thermodynamic process at different scales, through  
752 the variation of the polytropic exponent values. This fact would imply different  
753 efficiency values for the device at different scales. As a first approach, the results  
754 reveal that non-equilibrium states, which would be less evident in scaled model  
755 according to the sensibility of the polytropic exponent, would become more relevant  
756 as the scale is increased towards the size of the prototype.

757 The contribution of this research to the existing literature is to provide a better  
758 understanding of the initial scale effects studies, Weber [65], Falcão & Henriques [12],  
759 in the sense that reveals how the thermodynamic scaling requires a different adjustment  
760 as the standard scaling applied to other process involved in OWC performance. In fact,  
761 the accuracy of thermodynamic processes experimentally simulated increases downward  
762 —from full scale to model—, due to the minimization of transient states between equi-  
763 librium states, while the rest of process involved in OWC performance gain in accuracy  
764 upward —from model to full scale—.

765  
766 When balancing the scaling on both water side and air side involved in the OWC  
767 dimensional problem, there might be a counter-effect in several processes involved.  
768 Wave action, turbulence and hydrodynamic pressure is expected to be conditioned and  
769 somewhat limited at small scales when compared with full-scale performance. On the  
770 contrary, thermodynamic processes directly related with pneumatic performance and  
771 efficiency through polytropic process, might be disturbed at larger scales due to transient  
772 states required to reach an even distribution of thermodynamic variables over the entire  
773 system. This fact could indicate that devices smaller than full-scale plants built nowadays  
774 could have a better performance.

775  
776 The implementation of a numerical model to study scale effects is a matter of study  
777 with several difficulties in experience of the Authors of the proposal, Moñino *et al.*  
778 [47], Medina-Lopez *et al.* [40]. Even if many aspects can be successfully represented,  
779 namely the real gas performance, two-phase air and water model, wave impingement and  
780 radiation-diffraction through deformable mesh feature, etc., one of the main issues is the  
781 correct representation of the turbine. The free rotation of the turbine driven solely by  
782 the air phase displaced by the water phase inside the chamber is difficult to simulate.  
783 In fact, the common procedure is to impose a rotation to the turbine domain, which  
784 in turn affects the radiation-diffraction and pressure-air flow coupling, or to replace the  
785 turbine by an orifice or actuator disk model, Teixeira *et al.* [60], Moñino *et al.* [47]. In  
786 both cases, while the pressure peaks in compression and expansion can be successfully  
787 represented, all details regarding pressure-volume states through the polytropic process  
788 are not properly simulated. For that reason, to retrieve some additional information on  
789 scale effects seems to be less reliable than the information deduced from a comprehensive  
790 theoretical basis -even if some simplifications are assumed- to be later observed in an  
791 experimental model.

792  
793 In order to continue with this research, the next step would be to conduct some  
794 numerical simulations where all the effects indicated above would be analysed. Taking  
795 step further, it would be required an experimental study to reproduce the performance  
796 of a real-scaled OWC model in a laboratory and to check the results of this research and  
797 the numerical simulations. Conceptually, it seems relevant to identify a measure of the  
798 distance, in phase space, from the equilibrium condition of the transformations, in order  
799 to estimate its value in both the model and the prototype.

800 **Author contribution**

801 **Ángel Molina:** Analysis, writing & editing.

802 **Sandro Longo:** Analysis, concept, writing & editing.

803 **María Clavero:** Project leading, manuscript revision.

804 **Antonio Moñino:** Project leading, concept, writing & editing.

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971 **List of Figures**

972	1	Variation of the polytropic exponent with the non-dimensional volume of	
973		the system, according to the instability analysis indicated in eq.(30). . . .	18
974	2	Variation of the polytropic exponent with the non-dimensional volume of	
975		the system (according to eq.(30)). The shaded areas indicates the two	
976		cases of study. The blue one represent the full-scale device, and the red	
977		one the scaled model. . . . .	19
978	3	ressure obtained for the two cases studied. . . . .	20

979 **List of Symbols**

$A_c$ : Cross-section area of the air chamber – [m<sup>2</sup>]  
 $A_{pto}$ : Cross-section area of the PTO device – [m<sup>2</sup>]  
 $C_r$ : Coefficient of resistance – [–]  
 $C_y$ : Specific heat under constant variable  $y$  – [J/(mol K)]  
 $F_{aer}$ : Aerodynamic forces – [N]  
 $F_{in}$ : Inertial forces – [N]  
 $g$ : Gravity acceleration – [m/s<sup>2</sup>]  
 $h_0$ : Chamber height at rest – [m]  
 $K_1$ : Turbine damping coefficient – [kg/(m<sup>2</sup>s)]  
 $K_T$ : Isothermal compressibility coefficient – [Pa<sup>-1</sup>]  
 $Ma$ : Mach number – [–]  
 $m$ : Mass of air – [kg]  
 $m$ : Polytropic index – [–]  
 $n$ : Polytropic exponent – [–]  
 $p$ : Pressure – [Pa]  
 $p_{0c}$ : Pressure inside the chamber when  $\eta = 0$  – [Pa]  
 $p_a$ : Atmospheric pressure – [Pa]  
 $p_c$ : Air chamber pressure – [Pa]  
 $r_{(\dots)}$ : Ratio between the value of the variable (...) – [–]  
 $\tilde{p}_c$ : Relative pressure in the chamber – [Pa]  
 $Q_m$ : Mass flow – [kg/m<sup>3</sup>]  
 $R_0$ : Universal gas constant – [8.31 J/(K mol)]  
 $Re$ : Reynolds number – [–]  
 $S$ : Entropy – [J/K]  
 $T$ : Temperature – [K]  
 $t$ : Time – [s]  
 $V$ : Air volume – [m<sup>3</sup>]  
 $V_0$ : Initial air volume of the chamber – [m<sup>3</sup>]  
 $v$ : Space average air velocity – [m/s]  
 $v$ : Molar volume – [m<sup>3</sup>/mol]  
 $We$ : Weber number – [–]  
 $\ddot{x}$ : Acceleration – [m/s<sup>2</sup>]  
 $z$ : Vertical direction – [m]

Greek

$\gamma$ : Polytropic exponent for adiabatic process – [–]  
 $\eta$ : Instantaneous cross-average water level in the chamber – [m]  
 $\lambda$ : Length scale factor – [–]  
 $\nu$ : Kinematic viscosity – [m<sup>2</sup>/s]  
 $\rho_a$ : Air density – [kg/m<sup>3</sup>]  
 $\rho_c$ : Gas density inside the chamber – [kg/m<sup>3</sup>]  
 $\rho_w$ : Water density – [kg/m<sup>3</sup>]  
 $\Phi$ : Potential function – [m<sup>2</sup>/s]

980